Glenn Research Center

Altitude Wind Tunnel

Section 106 Check Sheets

Project, activity or undertaking:	FY2006 Demolition of Altitude Wind Tunnel (Bldg. 7)
Project Description and reasons for undertaking:	 The project will demolish the Altitude Wind Tunnel exterior portion of the Building 7 complex but retain the Test Section in Building 7. The demolition of a major structure will require the abatement of Lead Based Paint and Asbestos Containing Materials. The demolition will entail the removal of all of the structure of the AWT (approximately 3,000 Tons of Steel that will be recycled) and concrete foundations to two feet below grade. The AWT has not been used as a wind tunnel since 1960. The exterior of the tunnel is rusted and deteriorated past the point of repair. It is an eyesore and is becoming a safety hazard.
	Cost estimates have been prepared to perform lead based paint abatement, repainting of the exterior shell and structure, and some minor structural repairs. That work was estimated to cost \$4,000,000. This would only be a cosmetic remedy.
	The structure would serve no useful research or institutional purpose after that investment.
Ohio Historic Inventory form number	CUY-4587-15
Project manager or Point of Contact:	Robert Houk
Date:	July xx, 2006.
Date of GRC Evaluation Checklist/REC:	October 11, 2004
GRC Facility Preservation Officer:	Leslie Main

Yes. The demolition of the Altitude Wind Tunnel will be funded from the NASA Construction of Facilities (CoF) Program.
Yes.
 The Ohio Historic Preservation Office, the Western Reserve Historical Society, the Cleveland Landmarks Commission, NASA Retirees, the NASA History Office, the Cleveland Restoration Society, and the cities of Cleveland, Fairview Park, North Olmsted and Brook Park. There are no known Tribal Historic Preservation Offices (THPO) with interest in Lewis Field.
A Community Awareness meeting was held on April 27, 2006 on the GRC Campus. Meeting announcements were sent to local public libraries (Fairview Park, North Olmsted, Brookpark, Cleveland Public), the Sun Post/Sun Herald, and the Cleveland Plain Dealer. Announcements were sent to NASA retirees and an announcement was published in GRC's Aerospace Frontiers.
The meeting discussed the history of the AWT, the demolition process, the environmental impact, and the historical mitigation process. The meeting ended with a Question and Answer session.

Step 2. Identify Historic Properties	
Identify if Significant:	
A. Significant due to association with events	Yes. The AWT was significant in early development of jet engine aircraft. The AWT contributed to aeronautic research, aerospace propulsion and space flight systems.
	NASA Glenn Research Center has been at the forefront of jet and rocket engine studies in general, particularly under altitude conditions. The AWT was a primary component in the center's early research. The ability to test full-size engines instead of just a single cylinder, as was previously done, resulted in a more rapid transition from design to flight testing.
	The Altitude Wind Tunnel, which first began operation in 1944, was the first wind tunnel in the country capable of testing full-scale engines under simulated altitude conditions. The facility was converted from a wind tunnel to a vacuum chamber in 1962 and was renamed the Space Power Chamber. The new Space Power Chamber was utilized through the early 1970s for Centaur system testing and separation tests.
	The AWT was the nation's first wind tunnel capable of testing full- scale engines in simulated altitudes. Engine power, speed, drag, vibration, and cooling could all be analyzed in altitude conditions. The massive exhauster system and refrigeration system could simulate altitudes up 50,000 feet and produce speeds of 500mph. The tunnel's support buildings and systems were also used to operate the Icing Research Tunnel.

Wartime Research Almost all of the NACA's research during the war concentrated on improving existing military aircraft. A notable exception was the Altitude Wind Tunnel's very first test, the Bell YP-59A Airacomet. The Airacomet was powered by the GE I-16, which was the first turbojet engine manufactured in the United States. The I-16 was based on a Whittle jet engine, which was secretly brought from Britain to the United States in 1941. Despite the enhancements made in the Altitude Wind Tunnel, the Airacomet remained too problematic and inefficient to be used for World War II combat.
Much of the center's efforts during the war were concentrated on solving cooling problems for the new Boeing B-29 bomber's Wright R-3350 engines. The aircraft had been rushed through production for the war, but the R-3350s often burned up or failed at higher altitudes. The engines underwent a series of investigations over the course of six months in 1944. NACA researchers were able to resolve the cooling problem and increase fuel efficiency by 18%. These improvements were not integrated into the engine design until after the war. US Strategic Bombing Commander, Curtis LeMay, opted to forsake the high-altitude precision bombing for low-altitude incendiary bombing which caused more casualties but would not strain the B-29's engines.
Early turbojets Following the war, the NACA was finally able delve into new research, particularly with turbojet and ramjet engines. Turbojet studies actually began in late 1944 with studies on the Westinghouse 19B and 19XB engines, the GE TG-180 engine and afterburner, and the Lockheed YP-80A Shooting Star, which was the first jet aircraft

entirely manufactured in the United States. Air distribution, windmilling, and basic operating problems with the Shooting Star's GE I-40 engine were improved. The YP-80A evolved into the F-80 and was used extensively in the Korean War.
The Altitude Wind Tunnel underwent several upgrades in the late 1940s and early 1950s to increase the tunnel's speed and altitude capabilities. This included reducing the size of the test section. Originally entire fuselages could be inserted, but even with the smaller size test section, entire engines and nacelles could be used. Testing continued on ramjets, turbojets, and the jet-powered turboprops.
Mercury Program By the late 1950s, though, other more advanced facilities, including the new Propulsion Systems Laboratory, were relied on for more and more engine testing. As the space program emerged and the center became part of NASA, the facility began to be utilized for its cavernous space rather than its wind tunnel capabilities. Escape rocket tests for the Mercury capsule were performed inside the wide section of the tunnel. In 1959, the original astronaut corps traveled to Cleveland to be test their ability to bring a tumbling Mercury capsule under control in the Multi-Axis Spin-Test Inertia Facility Trainer apparatus which was also installed inside the Altitude Wind Tunnel.
Space Power Chamber Between 1958 and 1960, the lab refocused its efforts almost completely towards the space program. Although the AWT had already played a prominent role with the Mercury Program, its

	continued use was in question. Unwilling to let this historically significant facility fall into obsolescence, NASA Lewis administrators decided to convert the tunnel into vacuum chamber.
	During this conversion process, two massive bulkheads were installed inside the tunnel, creating two vacuum chambers. It also included rewelding the tunnel's joints, installing a new vacuum pump house, and creating a dome that could be removed to insert test equipment. On September 12, 1962 the Altitude Wind Tunnel was officially renamed the Space Power Chamber or SPC.
	Centaur In 1961, Center Director Abe Silverstein convinced NASA to transfer the Centaur Program from Marshall Space Flight Center to NASA Lewis. A mock-up Centaur rocket was installed in the Space Power Chamber to study entire systems tests in a space environment. The other end of the tunnel was used for shroud separation tests. These tests contributed to the Surveyor, Orbiting Astronomical Observatory, and many other Centaur missions.
	Rehab By the mid-1970s fewer tests were conducted in the Space Power Chamber and it fell into obsolescence. There was a proposal in the early 1980s to overhaul the facility and convert it back into a wind tunnel for icing research tests. After several years of preliminary studies, the \$150 million renovation was cancelled and the facility has remained unused.
B. Significant due to association with persons	No. Minor association with John Glenn and the Mercury Program Astronaut Corps, but not enough to qualify for this criteria.
C. Significant because it embodies distinctive	No. The AWT was the first wind tunnel to be able to simulate

characteristics	atmospheric conditions, such as temperature and air pressure, for conditions above sea level. Also, the related Refrigeration Building 9 is significant due to the engineering feat to achieve this large scale refrigeration system. (Bldg. 9 is in use, directly supporting another research facility, the Icing Research Tunnel {IRT}. The IRT is the currently the most utilized wind tunnel at GRC.) However, the AWT wind tunnel, chambers, and supporting facilities are typical of similar research facilities, and are not unique.
D. May yield information important in prehistory or history.	No. The site was significantly disturbed during the construction of the AWT and its supporting facilities and utilities. Because of this, the site would not reveal any information or artifacts.
Identify if it has Integrity:	
Property is able to be preserved; has physical integrity to still communicate what made it significant. Contains 1 of the 7 aspects of integrity: location, design, setting, material, workmanship, feeling, or association.	The property retains much of its original integrity, but the structure is deteriorating due to corrosion. The facility could not be used again as a wind tunnel without significant investment. And if would only duplicate facilities that are currently operational. All of the vacuum systems have been removed from the Space Power Chamber and larger vacuum facilities exist at GRC that can be used for testing space craft. The annual maintenance costs for AWT are \$93,000. And as stated above, a major project to stabilize the facility would cost \$4,000,000 (see Appendix B for the cost estimate) and would not provide a productive research facility.
Determination of Undertaking's Area of Potential	Effects (APE)
Alternative locations?	Not applicable. The former capabilities of the AWT are duplicated or exceeded at other research facilities either at Glenn Research

	Center or two other NASA centers, Langley Research Center or Ames Research Center.
	The cost to relocate the AWT or build a duplicate facility would be cost prohibitive.
Disturbance of the ground?	Disturbance expected during demolition. The massive concrete support structures and foundations will be removed to two feet below grade. The area will be repayed for stormwater control. The paving will be adequate as a temporary parking area.
Locations from which visible?	The AWT is visible from GRC campus and the top part of the tunnel is visible from Cleveland Hopkins Airport. See the attached photographs.
Change in land use, traffic, public access, etc.?	Initially, the site will likely be used for parking. There are discussions about building a new Refrigeration Building on the site to enhance the capabilities of the IRT or building a new research facility at that site.
Cuth an information and to ADE	
Gather information on the APE See Appendix A OHI forms which include a site pla	an and building plan and Appendix C. Photographs of the Facility
See Appendix A, OHI forms, which include a site plan and building plan, and Appendix C, Photographs of the Facilit	
Identify historic properties within the APE	
The property is located within an area that the Gray & Pape 2002 Survey identified as eligible for listing as a Historic District. The property is in the vicinity of the Zero Gravity NHL. The property is very close to the Icing Research Tunnel, a facility that may be eligible for NRHP listing.	
<i>Evaluate Historic Significance of Property; NASA Determination of Eligible or Ineligible</i> Eligible. Continue to Step 3 of Section 106 process The Altitude Wind Tunnel complex is eligible for listing on the	
Englote. Continue to Step 5 of Section 100 process	National Registry of Historic Places (NRHP) because of its legacy
	of contributions to Aeronautics and the early space program, especially the Mercury and the Centaur programs as described in

	Step 2a.
	The most significant part of the AWT is the test section. The large structure of the AWT and its supporting technology were only there to create test conditions of full sized and scaled test articles. As part of this demolition project, NASA will be saving the test section of the AWT and will make it into a Historical interpretation center. Even though the AWT control room has been gutted, it will also be included as part of the interpretation center.
Step 3. Assessing Adverse Effects	
Finding of Adverse Effect	
0 0	IP, NASA finds that the demolition of this property meets the criteria val of the tunnel structure and foundations, but the Test Section of the
Step 4. Resolving Adverse Effects	
Avoiding Adverse Effects	
A. Moving the undertaking to an alternate site	This alternative would be cost prohibitive. There are facilities at GRC and within other NASA Centers that duplicate the capabilities of the AWT and the Space Power Chamber.
B. Using an alternative design	There was a proposal in the early 1980s to rehabilitate the facility and convert it back into a wind tunnel for icing research tests. After several years of preliminary studies, the \$150 million renovation was scrapped and the facility has remained unused.

C. Pursuing an alternative to the undertaking 1. After the AWT was no longer needed as a wind tunnel, the structure was adapted three times. The first adaptation was to test the escape rocket for the Mercury capsule. Scale models of the Mercury capsules were tested inside the wide section of the tunnel. The second adaptation was in 1959, when the original astronaut corps traveled to Cleveland to be test their ability to bring a tumbling Mercury capsule under control in the Multi-Axis Spin-Test Inertia Facility Trainer apparatus, which was also installed inside the Altitude Wind Tunnel. The third adaptation was changing the structure into the Space Chamber (see Step 2a Subtitle Space Power **Chamber**). Since then, NASA has built newer facilities, at GRC and other NASA Centers, that perform the same functions but have expanded the capability beyond what could be accomplished even if the AWT was rehabilitated. 2. Rehabilitation of the exterior structure only as an alternative was considered. The work would involve removing rust, replacing corroded steel panels, preparing the metals, mitigating contaminants (i.e., abating lead based paint and asbestos containing materials) and painting the structure. A construction cost estimate was prepared by Crawford Consulting in August 25, 2003. Estimate for the painting was \$4,000,000 (approximately 20% less than the cost of the undertaking). If this alternative was pursued, NASA would have the recurring costs of repainting the structure every 8 to 10 years. That would be a significant cost for a facility that is no longer serviceable or of use to the Government. 3. Rehabilitation of the structure for a new use is an alternative that was considered (see Step 4b). NASA has chosen the approach of not doing anything over the past D. No undertaking at all.

	20+ years. As a good steward of government assets, with the increasing awareness of Safety and Environmental concerns of the deteriorating structure, NASA believes that something must be done within the next few years.
Mitigation Measures	
• Alternative design & limiting the magnitude of the project.	NASA considered the demolition of the whole AWT. NASA has decided to retain the AWT Test Section, improve the area of the Test Section and renovate the former AWT Control Room as an historic interpretation area. Displays of photographs of testing within the AWT, as well as other interpretive materials, will be set up. The AWT Test Section will be a tour stop for guided tours of Lewis Field.
Alternative location	There are no alternative locations.
Rehabilitating some historic properties	Adjacent to the AWT is the Icing Research Tunnel (IRT). The two facilities were built during the Second World War. Both facilities were provided services by the Refrigeration Building, which will remain in service after the demolition of AWT is complete. The IRT is already considered a national landmark by the American Society of Mechanical Engineers (ASME) and may be eligible for listing on the NRHP.
Planning for preservation and maintenance	Please reference Step 4.c2.
• Moving historic properties or marketing the property for donation, sale, or lease	Due to the size and obsolescence of the structure, this alternative would not be feasible.
 Documenting property before destroying it • 	This is significant because the technology tested and developed by NACA and NASA in the AWT and the Space Power Chamber are more important than the structure. Properly compiling historical

scientific	c and engineering documents is important to NASA.
Because	of this, NASA is planning the following Historic
Mitigatio	on and Documentation for the AWT;
1)	Retaining the Test Section and Control Room and setting
	it up as interpretive site and tour stop. NASA will
	produce museum quality display boards the show the
	history of the AWT and the technology that was
	developed from the testing performed there.
2)	A Community Awareness meeting was held on April 27,
	2006 on the GRC Campus. Meeting announcements were
	sent to local public libraries (Fairview Park, North
	Olmsted, Brookpark, Cleveland Public), the Sun Post/Sun
	Herald, and the Cleveland Plain Dealer. Announcements
	were sent to NASA retirees and an announcement was
	published in GRC's Aerospace Frontiers. The meeting
	discussed the history of the AWT, the demolition process,
	the environmental impact, and the historical mitigation
	process. The meeting ended with a Question and Answer
	session. A monograph will be published recording the
	history of the AWT. The monograph will include full-
	sized photographs.
3)	A web site with public access will be developed for the
	AWT. Historic photographs of the construction and
	testing within the Test Section, the tunnel, and in the
	Space Power Chamber will be available for viewing.
	Photographs of the current state of the AWT and
	photographs documenting the demolition of the AWT will
	also be available for viewing. The text from the
	monograph will also be available for viewing.

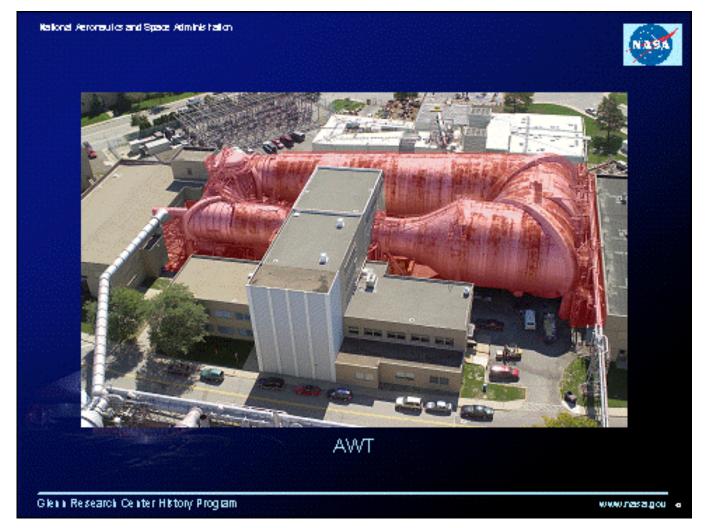
 4) HAEB/HAERS documents of the AWT complex will be prepared and archived. These documents will summarize the construction, historical context, technological significance, and a physical description of the AWT. Included in this documentation will be selected photographs and architectural drawings from NASA's files.
5) NASA will collect, appraise, and maintain a collection of historically significant documents that will become a permanent record of the AWT. These documents may include correspondence, architectural drawings, maps, scientific or engineering publications, and related materials.
 6) NASA will update the photographic images by digitizing unscanned negatives and photographs and uploading them to the GRC Imagenet database. NASA will perform several 360 degree images of the AWT before demolition begins. NASA will compile film and video of tests performed in the AWT and have the film/video digitized. From the above digitized files, NASA will produce a CD-ROM or DVD that will include photographs, panoramic photographs, video clips, and scanned documents. This disc could supplement the monograph or be distributed separately.
7) Oral interviews will be conducted with NASA retirees, facility and program managers, and others. These interviews will be recorded and transcribed. Selected interviews will be videotaped.
8) NASA will produce a documentary video that would describe the facility, its history, and research programs.

Recovering data from archeological site	See Step 2d.			
Accepting loss of historic property	NASA is willing to accept the loss of this property.			
	NASA Headquarters has concurred with and advocates the proposed demolition of AWT. With the Aeronautics Research budget being 1/17th of NASA's overall budget, funding for improving Aeronautics research facilities at GRC would be better invested in other wind tunnels (IRT, the 10x10 SWT, the 8x6 SWT) or the PSL #3 & #4.			
	The facility, as configured, has been out of service for more than 30 years. During the 1960's, the tunnel underwent major modification to create the Space Power Chamber to support specific test goals for the Centaur Program. Since then, no significant research work has been done in the tunnel circuit since Centaur. Because the AWT is a large structure, the maintenance costs for the facility are very high and the facility is in poor condition.			
	During the 30 years that the AWT has been out of service, significant hardware has been removed from the Wind Tunnel complex such as the fan, drive shaft, turning vanes, and compressors. All of the vacuum pumps for the Space Power Chamber have been removed.			
	The Government completed a preliminary engineering study in the mid-1980's to substantially modify the AWT circuit for use as a new Icing Research Tunnel Facility. The project was cancelled due to the excessive costs involved to rehabilitate the AWT.			

	There are no current NASA mission requirements for either near- term or long-term use of the tunnel circuit other than as a potential Visitor's Center Tour stop. A recently completed cost estimate to do minor exterior repairs and repaint the tunnel circuit and utilities was in excess of \$4.0M. See appendix B. Currently, NASA GRC is a land locked site. The area that will be cleared by the demolition of the AWT will allow NASA to locate new research capability in that area. The current NASA GRC proposal is to remove the entire AWT circuit except for the Test Section within the High Bay Building as part of the Bldg. 7 complex. The only major costs involved with retaining the Test Section would be for closures on both sides of the High Bay and residing the High Bay after existing asbestos panels are removed.
Notify council – participate in consultation? Consulting parties involved in discussion Public comment & involvement during resolution of adverse effects	Yes. NASA will invite the ACHP to consult in this process.
MOA	

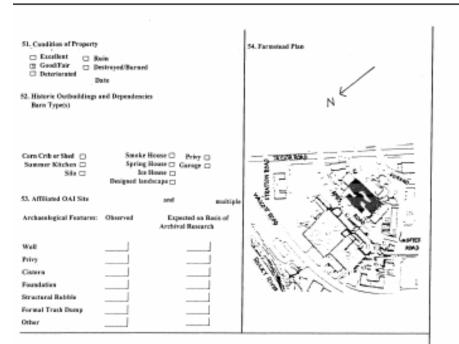
Appendix A

OHI Forms for the Altitude Wind Tunnel



Extent of Demolition for the Altitude Wind Tunnel (Highlighted in Red)

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42. (Caufd)

In 1995, a one-story, three by one bay, 3,000 square fact addition was constructed on the north facule of the west "wing" of Building 1. This addition matches the original building in construction and cauterials.

The bays of the two-mory wing are delineated by paired and single plate gloss piveting windows with continuous concrete allb and spars of alternating projecting courses of brick between each set. These windows replace the original grouped set of harizontally-divided, wait-panel stall said windows. The main entry into Building 7 is from Amers Road (north) through a set of deable glass doors with a glass transion and box (conf.d).

43. (Cont'il)

while also advancing technologies in acrospace propulsion, and space flight systems.

The Altitude Wind Turnel (AWT) has been cited as historically the most important facility at the LaRC. The AWT was designed as a closed circuit turnel with a JP discreter fan oppfile of producing air velocity as high as 425 mph at simulated altitudes of 30,000 fact and at low as 258 mph at 1,000 fact. The U.S. Army presend for the design and construction of the AWT in 1942 because of 16,000 fact and at low as the analyze and solve problems of adequate engine cooling. Private indextry could not affind such a large and expressive facility, so it was 100 to NACA to delive into solving such equine problems and, in design such causes low-lead in development.

Storl shortages slewed construction of the added-steel shell of the tunnel. Design was slowed also by the requirement for an extensive refrigention system to save both the AWT and the king Research Tunnel, also is nonstruction. The Certier Corporation was brought in to design this system, which broke ground in terms of Targe scale engineering fease and contributed, through testing performed with its help, to (cont/t).

44. (Cont'd)

located to the year of the shop and office building. Parking is allowed in the payed area underseath the tunnel when texts are not being run.

Continuation Parts One

3. Photographs previded by NASA: C-3992, AERL 4804, AERL 5064-A, C-5681, C-5308, C-8983, C-19794

28. Sam W. Emerson Co., Cleveland, Ohie: The Carrier Company,

34. Building Dimensions: one wing measures 74 x 23.

42. metal campy above. A similar analy existed in the seventh bay (see the west "wing" of the office section), but the construction of the one-story addition filled in this space. A secondary antry is located on the west facule of the two-story wing and consists of paired glass and metal doors, a glass transom and a box matal canopy. A narrow overhead door is also located on this facade,

The tail section of the T-shaped building contains a shop area in its northern and and a hatch into the tannel on its southern end. The northern and of this tawar intersects with the office/shop wing. Its north facade faces onto Ames Road. This facade costains a large, eventual dear, which leads into an open shop area. The exterior of this four-story tower is clad with tan brick as its numbers and and is partially clad with horizontal metal siding on its southern end. The northern section formerly held horizontal ribbon, windows on the gave apper floors. These windows have been filled in with brick, but the constrate sills are still visible. To the rear of the tower, some of the original barizontally divided multi-paned windows have been covered over with metal siding.

The wind named intersects the tower on its southern and where one of the text sections is located. The tunnel crasses a restangular loop that is larger on its wamen and than on its canters and. The named structure is supported by large concrete piers and steel supports. The interior of the turnel, designed as a closed circuit system, has been madified savaral times over the years. In 1962, the facility was modified as the Space Power Laboratory to allow for environment testing of the Atlas/Censuar relation. The latest modifications, designed to allow research nests on ising, propellor-powered and WSTOL vehicles, were not successful. In 1991, the name of the facility was changed to the Microwave Systems Laboratory.

Bailding 78

Building 78, constructed hereven 1951 and 1952, is positioned cosswipt understath the custom and of the wind tunnel loop. It is a one-story, flat-cooled structure with a basement and measures 47 by 28. The building has a concrete foundation and is clud with tan brick. The main access into Building 78 is through a metal door on the cast side. A metal, bax canopy is located above this entry. Access doors are also located on the south and west facades. The west entry consists of a rectal door with metal side parals flanked by horizontally-divided. multi-gamed windows. A verted transom is located above. The south easy consists simply of a pair of metal doors. The north facade has a large opening filled-in with glass block. An arcaway on this facade is open to the basement level with a metal pipe railing surrounding the opening. The costor pit of the wind named is located directly under the most leg of the loop and is adjacent to the north end of Hallding 78. It is a concrete structure with metal cladding. Originally, this building served as the Water Pump House for the Altitude Wind Tunnal. It was expipped with four Pairbanks-Moree 250 horsepower pumps. Around 1964, two of dasse pumps were remeved for use at the Plum Brook Station. Also is the 1960s, the facility was modified as the Solar Mirror Cleaning Solar Pawer Laboratory Parap Hease, which involved the Installation of four cleaning tanks and new sentilation and plumbing systems.

43. shortoning the war. Carrier built and tested many original components in its design for the refrigeration system. The refrigeration plant contained 14 Carrier contribution compressors and a unique heat exchanger capable of producing a minimum temperature of +48 F. The new compressor developed in this endets/or became one of the company's standard products after the war.

The first anofficial text in the AWT was performed in February 1944. Although the tunnel had been constructed with piston engine tests in mind, the first engine to be tested was the I-16 turbojet, which had been scenetly designed by the General Electric Company. For the test, an antiau Resultage of a Boll Aircraft P-38A with its wings sowed off was squeezed into the 20' diameter test section. The first official tests in the AWT, run in May, were conducted on the Wright R-3350 pixton origine, which was used in the B-09 Superfortness used in strategie bombing of Japan from the China mainland.

After the war, the AWT was adapted so test early tarbojet and tarbojetop engines at simulated altitude conditions. With the NASA change in mission, the AWT was converted to a vacuum facility to test tookets in 1958. In the early 1968s, the "Space Pawer Chamber" was nied to test the Centaur rocket, the important upper stage rocket facted by liquid hydrogen

45. Overall Cultural Resource Reconstationance Survey of NASA Lewis Research Center, Cleveland, Ohio, Gray & Pape, Inc., 1996.







Continuation Page One

3. Photographs provided by NASA: C-3992, AERL 4804, AERL 5064-A, C-5681, C-5508, C-8983, C-19794

28. Sam W. Emerson Co., Cleveland, Ohie: The Carrier Company

34. Building Dimensions: one wing measures 74 x 237.

42. must a campy above. A similar analy axisted in the seventh bay (on the verse "wing" of the office section), but the construction of the encourse addition filed in this space. A secondary analy is located on the west facede of the two-story wing and consists of patient glass; and rest doors, a glass transmit addition, a glass transmit and the west and analys. A cancer occurrent add to be located on this facede.

The tail section of the T-shaped building contains a shap area in its northarm and and a hatch into the tawnel on its southern end. The narthern and of this tawar intersects with the office/shop wing. Its north theade faces onto Armas Rand. This faced a contains a large, eventhead does, which hands its one an open shop area. The electricies of this free-sterp tweer is clad with taw brick as its narthern end and is partially clad with borizontal metal skilog on its southern end. The northern section flowner/y held horizontal risks with a southern end. The northern section flowner/y held horizontal risks with a southern end. The northern section flowner/y held horizontal risks with a southern end. The northern section flowner/y held horizontal risks are of the spiral brief. These windows have been filled in with brick, but he constrait sills or oil it within. To five rear of the tawer, some of the migrant brief contails divided multi-parts.

The wind named intersects the tower on its southern end where one of the text sections is located. The named creates a sectangular loop that is larger on its warmen and that on its southern and. The named structure is supported by large correct piers and need supports. The interior of the sumed, designed as a closed circuit system, has been maddied asympticity with years. In 1964, the field ty was medified as the Space Power Laboratory to allow for environment testing of the Atha-Centum relation. The latest medifications, designed to allow execution inters on ising, populate-powered and WSTOL, whiches, were not successful. In 1991, the name of the facility was changed to the Microware Systems Laboratory.

Bailding 78

Bialding 78, construented between 1951 and 1952, is positioned (crosswise undersmith the cuters or of if he wind hanned loop. It is a one-story, flat-tooffed structure with a basement and measures 47 by 28. The building has a concrete lowedness multi-clad with the hick. The main access into Bialding 78 is through a metal door on the cut tide. A metal, how campy is located above the interfaalso located on the routh and west flatades. The west entry consists of a metal door with result is deep partic flatade by barizontal by-divided, multi-particle with glass block. An uncreasing on this facade is open to the fast with result is deep. The need particle doors the facade by barizontal by-divided, multi-particle with glass block. An uncreasing on this facade is open to the loop and has need pipe reling surrounding the opening. The coster pit of the wind transmit is located dividy under the test log of the loop and has algoen to the north call of Balting 78. It is to a concrete work metal challing. Chipking this building group of a pit of the loop and has algoen to the north of all dividing 78. It is a concrete work metal challing. Chipking this building group of a bia far far Party House for the Allinde Wind Tamal. It is van exployed with four Phithatis-Merse 201 hortspower pumps. Around 1964, two of these paraget were constant of the use at the Plane Brook Station. Also in the 1960s, the Flow work plants were related with were thereing Stellar Plane Internet for use at the Plane Brook Station. Also in the 1960s, which involved the low with and plantspin systems.

43. shortening the use. Carrier bulk and tested many original components in its design for the refugention system. The refrigension plant contained 14 Carrier contributed components and a unique loar avaluation random leads in the inferent temperature of -48 F. The new compression developed in this reflector became one of the company's tanabad postator after the set.

The first anofficial tast in the AWT was performed in February 1944. Although the tunnel had been constructed with pisten engine tasts in trivit, the first engine to be betted was the I-16 herbojet, which had been secretly designed by the General Electric Company. For the test, an entire final age of a Bell Alexand P-48A with its wings saved off was squeezed into the 20° diameter tast section. The first official tests in the AWT, run in May, were conducted on the Wright R-3330 piston ougline, which was used in the B-29 Superformers used in similarity benefiting of Japan from the China mainfield.

After the war, the AWT near adapted to test early ratiogist and tasbuprop engines at simulated admitude conditions. With the NRSA change in minimum, the AWT near converted to a secure facility to test sockate in 1958. In the early 1968s, the "Space Prover Chamber" was used to test the Centur reduct, the important upper stage socket facility by admitude hydrogan.

45. Overall Cultural Resource Recontraissance Survey of NASA Lewis Research Center, Cleveland, Ohio, Gray & Pape, Inc., 1996

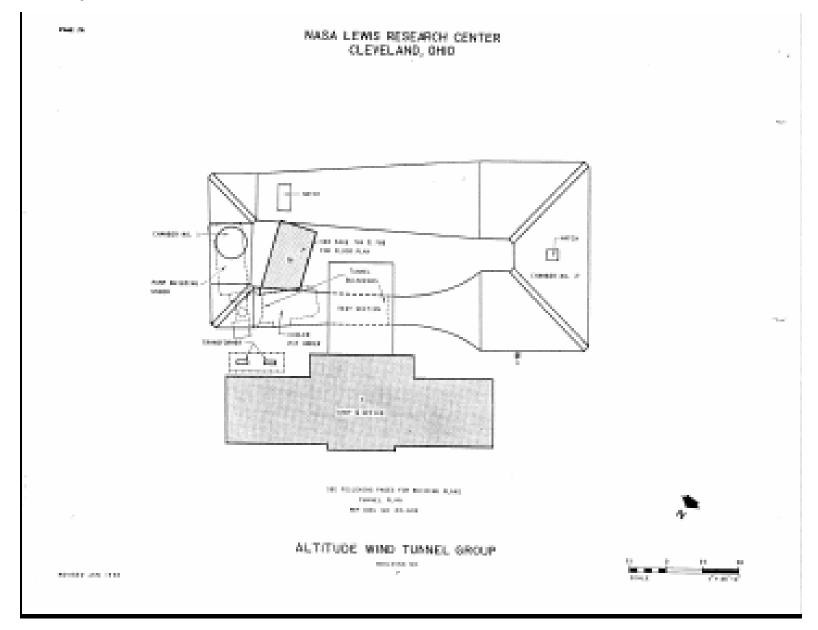




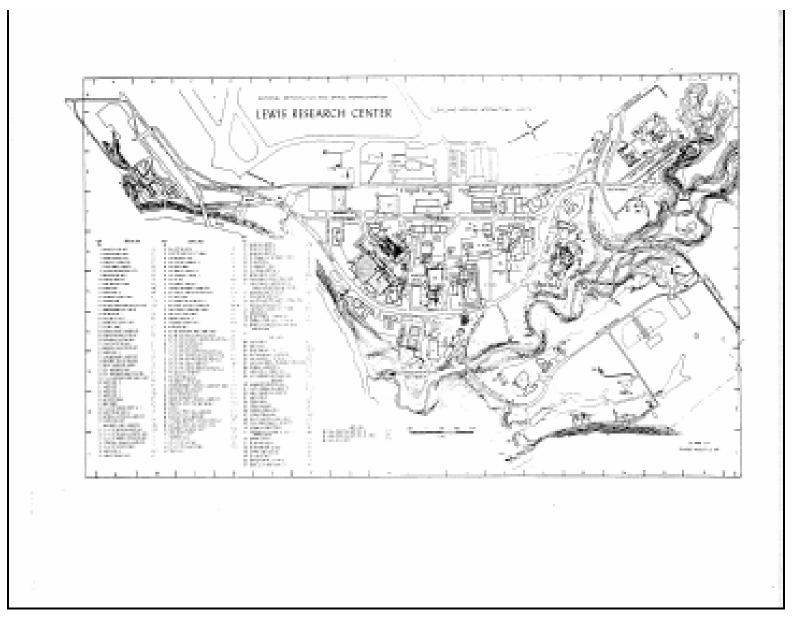
Yound Nagros

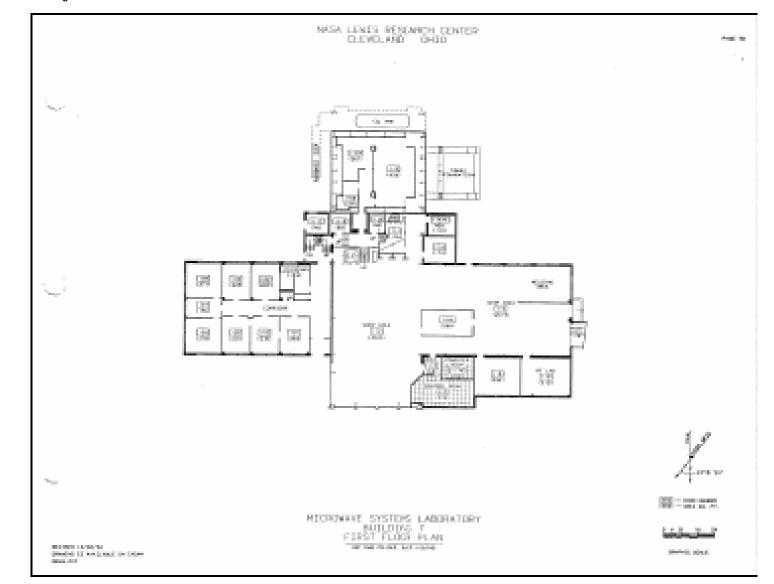


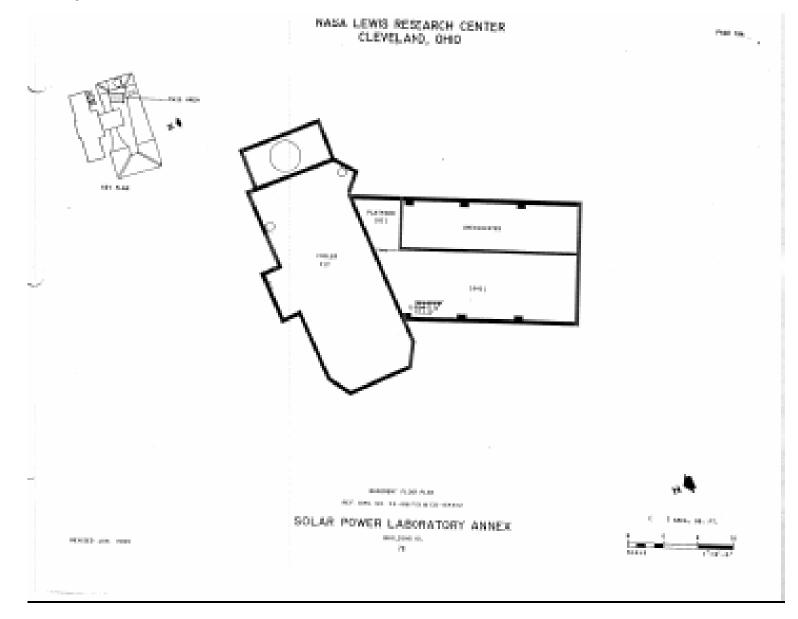
Recordation of the Glenn Research Center – Section 106 Process

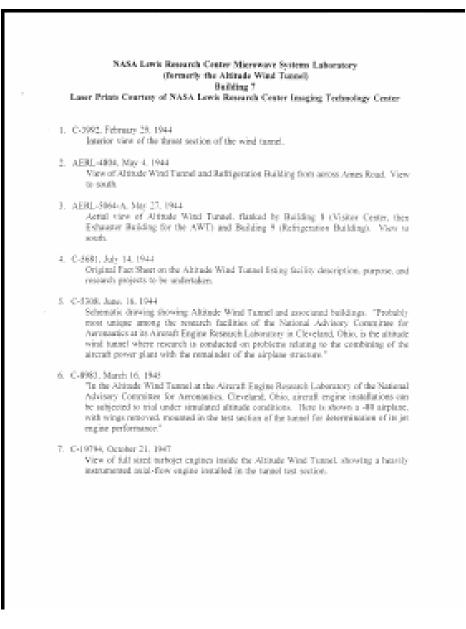




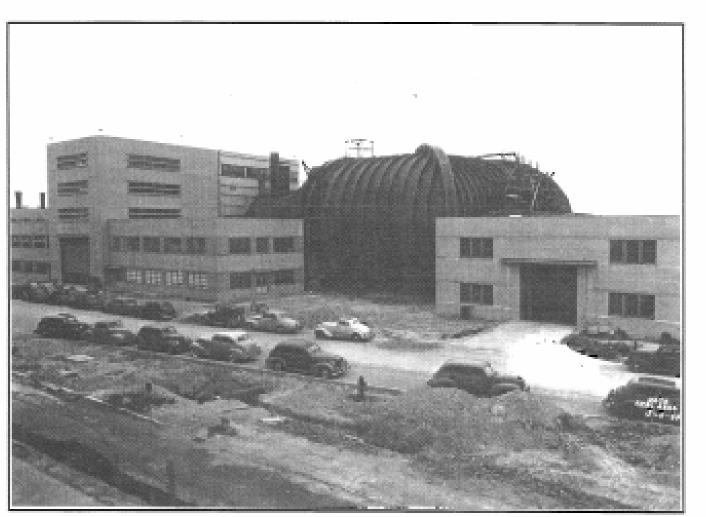




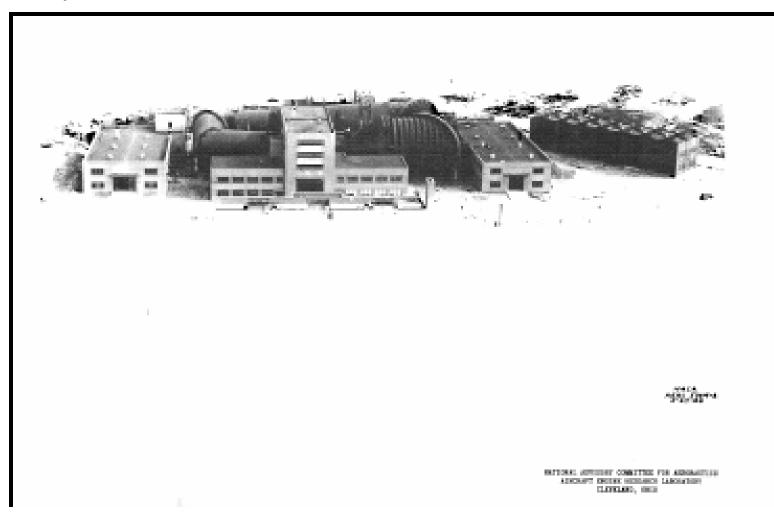


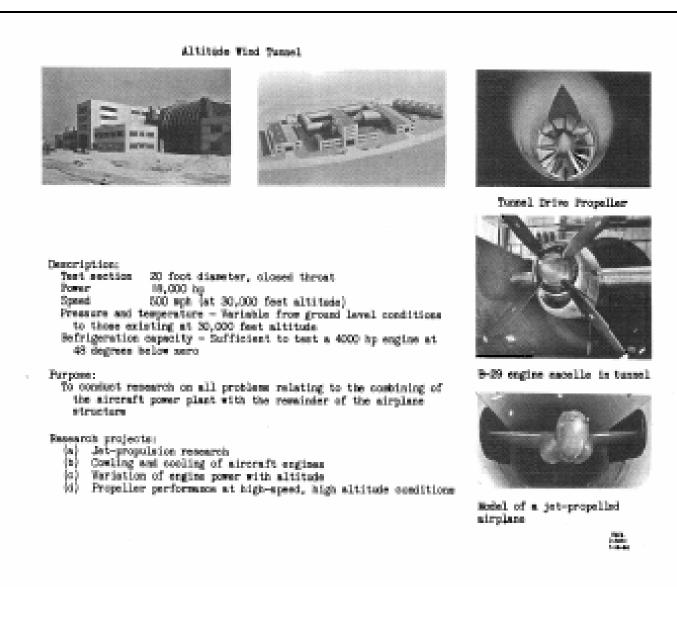


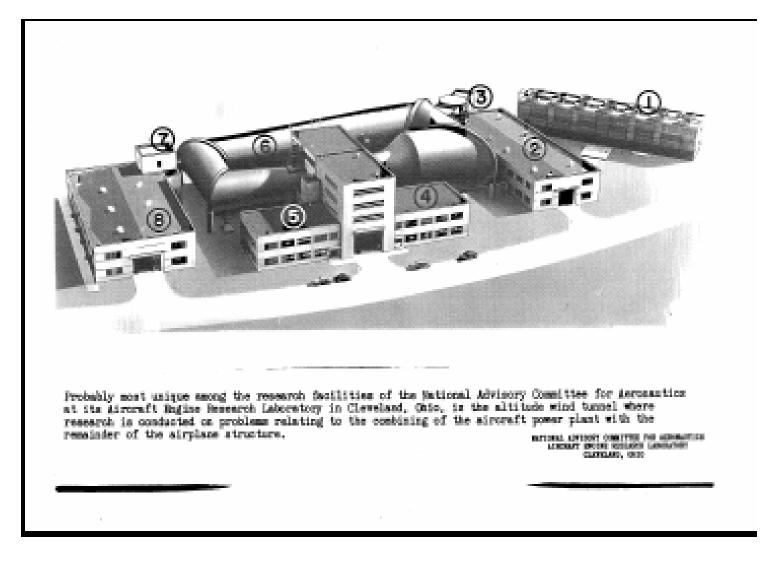


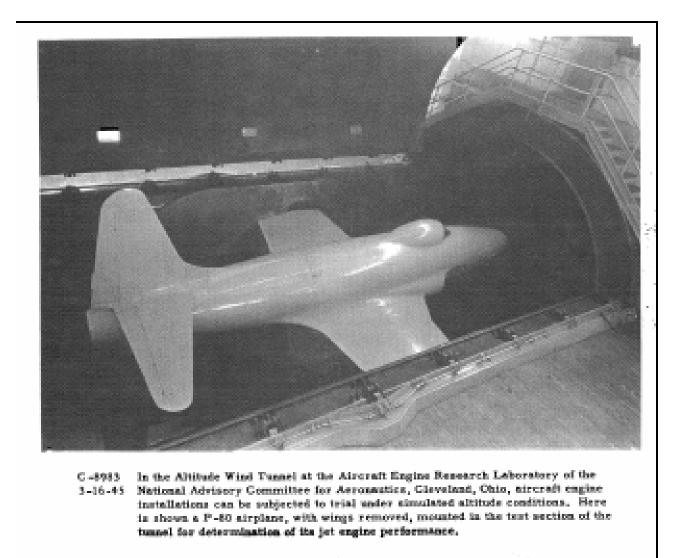


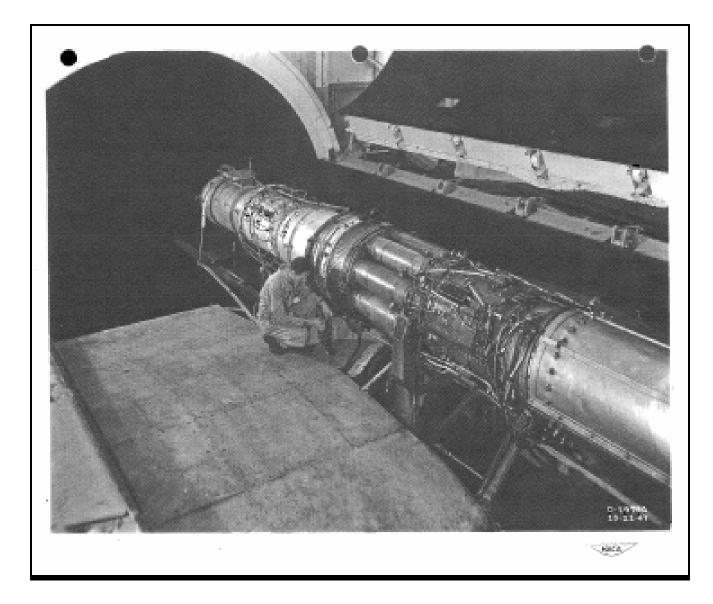
A 1944 photograph of the Aldude Wind Turnel.











Appendix B

Cost Estimate for Repairs and Painting of Altitude Wind Tunnel Structure

NOTES / CLARIFICATIONS / QUALIFICATIONS Project: NASA Glenn Research Ctr. - Building 7 Painting Only Location: Cleveland, Ohio Job Number: 2003-24 Note Number Description This estimate was prepared to help the National Aeronautics and Space Administration (NASA) allocate funds for projects. 1 This report is intended to help / assist a knowledgeable systems analyst to decide which repairs or demolition projects can be accomplished with 2 present allocations or to request the proper amount of allocations for future years. 3 Caution is advised when choosing "selected" or "single" items because cartain tasks cannot be done without cost ramifications. Certain scopes of work must be combined to achieve the correct cost for the desired task. For example, the base bid is the basic portion of this project, and the other options must be associated with the base bid per each scope of work. 4 This estimate may also be used as a guide for determining future allocations or future repairs. If the construction / demolition is performed in smaller quantities / amounts, a premium must be added to the task. This estimate is calculated in 2003 dollars for the present year. To meet. your future year allocations, a 3% escalation fee per year should be added. On a majority of the projects, we have added two years of escalation. Particular attention should be used in all items listed as being self performed by NASA. This would include removing all loose furniture and 5 equipment before demolition occurs. If singular scopes of work are chosen, the general conditions may not be a true reflection of actual costs. A percentage can not be applied and 6 be accurate. Our firm used the resources of Precision Environmental Co. and Fuller Environmental Co. for work associated with the hazardous material 7 report. This includes asbestos abatement and containment and mercury clean-up... 8 Our firm has assumed that all crushed concrete materials stay on site and be used as backfill material at a later date or as needed. This can be incorporated in the process of bringing the demolished building back to existing elevations/grade or by hauling to the abandoned water retention basins at the edge of the site. 2 We have not allowed for any premium time to complete the work. Our assumption is that it is up to the contractor to determine their own work 9 schedule to best suit their needs. If a compressed schedule is required overtime premiums should be applied. Each building was priced as a stand alone building; therefore, giving the owner (N.A.S.A.) the ability to choose which year the demolition will 10 occur. We have assumed that the painting process would be performed one section at a time; therefore, the contractor would not be performing any 11 work during the winter months.

Crawford Consulting Services, Inc.		National Aeronautics & Space Administration	8/25/20	3 11:25AM		
					Total	
		Item	Takeoff		Unit	Grand
Location	CSI	Description	Quantity	Unit	Price	Total
		01 General Conditions		-		
01 General		01 General				
	01	General Requirements				
	01300	Administrative Requirements				
01 General		Field personnel clerk		week	400.00	20,800
01 General		Field personnel, field engineer		week	1,005.00	52,260
01 General	01310700	Field personnel, general purpose laborer	52.0	week	925.00	48,100
01 General	01310700	Field personnel, project manager	52.0		1,605.00	83,460
01 General	01310700	Field personnel, superintendent	52.0		1,510.00	78,520
01 General		Field personnel, timekeeper		week	780.00	40,560
		certain projects it it difficult for engineers, managers, or superintendents to trac	k man hours o	n jobs		
with lar	rge labor force ma	aking it necessary for a general contractor to use a timekeeper.				

01 General		Field personnel, safety person		week days	1,510.00	7B,520 63,700
and pr	the fact that this		ogress			
		Administrative Requirements Tota	al			465,92
	01400	Quality Requirements				
01 General	01450500	Testing	1.0	proj	25,000.00	25,000
		Quality Requirements Tota	al			25,000
	01500	Temporary Facilities & Controls				
01 General	01520500	Office trailer	12.0	mnth	650.00	7,80
01 General	01520550	Field office expense, office equipment	12.0	moth	250.00	3,00
01 General		Field office expense, office supplies	12.0	mnth	150.00	1,80
01 General	01520550	Field office expense, telephone bill	12.0	mnth	400.00	4,80
01 General	01520550	Field office expense, field office lights & HVAC	12.0	mnth	150.00	1,80
01 General	01530700	Safety	12.0	mnth	2,500.00	30,00
01 General		Temporary Enclosure	12.0	mnth	25,000.00	300,00
	a Vented Tempora	ary Enclosure Similar to Bridge Painting to Encapsulate Hazardous Materials & C	Overspray.			
01 General		Scaffolding	200.0	Csf	125.00	25,00
		as out of the safe reach of laborers.				
		1				

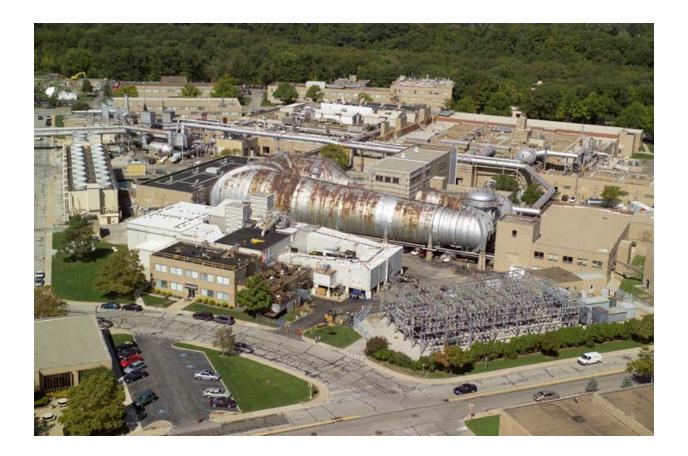
Crawford Consulting Serv	1088, INC.	National Aeronautics & Space Administration	8/25/2	003 11:25AM		
Location	CSI	Item Description	Takeoff Quantity	Takeofi Unit	Total Unit Price	Grand Total
01 General	01560100		1,500.0	Inft	45.00	67,500
Jersey bai 01 General	01580700	k fence, or orange plastic protective fencing. Signage Temporary Facilities & Controls Total	50.0	each	100.00	5,000 446,700
		General Requirements Total				937,620
01 General		01 General Total				937,620
		01 General Conditions Total				937,620
01 General		02 Demolition 01 General Site Construction Site Remediation				
01 General	02110300	Hazardous Materials Site Remediation Total	1.0	lsum	500,000.00	500,000 500,000
Our visua could be a	02220100 prep the metal inspection of applied.	Site Preparation Surface Preparation of Exterior Metal Skin of Wind Tunnel surface to receive paint. the exterior surface of the wind tunnel revealed a large amount of rusted areas, t	75,317.0 his would hav		12.50 corrected before	941,463 the paint
This cour	range from s	andbalsting to replacement of some of the exterior metal skin. Site Preparation Total				941,463
		Site Construction Total		·		1,441,463
01 General		, 01 General Total				1,441,463
1		02 Demolition Total			and the second second second second	1,441,463
		03 Painting				

Crawford Consulting Services, Inc.	National Aeronautics & Space Administration				8/25/2003 11:25AM		
Location CS	Item 8 Description		Takeoff Quantity		Total Unit Price	Grand Total	
	01 General 9 Finishes 0 Paints & Coatings						
01 General 0991062 Includes minor sand b	0 Painting of Wind Tunnel lasting, scraping, cleaning, and painting		75,317.0	sqft	10.00	753,170	
01 General 0991062	 Miscellaneous Painting walk and other micellaneous steel, and concrete foundation. 		1.0	lsum	200,000.00	200,000	
	Paints & Coati	ings Total				953,170	
	Finis	hes Total				953,170	
01 General	01 Gen	eral Tótal				953,170	
	03 Paint	ing Total				953,170	

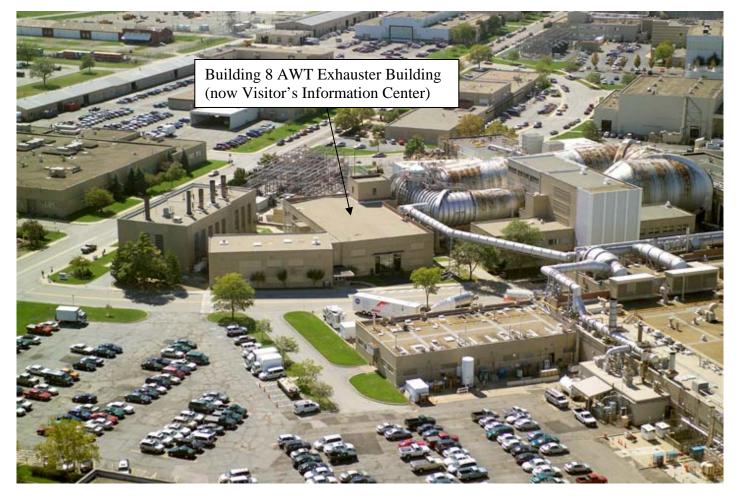
Crawford Consul	onsulting Services, Inc. National Aeronautics & Space Administration					8/2	25/2003 11:25AM	
·····	Division	Labor	Mat	Subs	Equip	Other	User	Total
01 02 09	General Requirements Site Construction Finishes	677,050 456,695 789,181	235,570 163,989	25,000 500,000	484,767			937,620 1,441,463 953,170
					; е	Contractor's Bonds & C scalation - 3% per y Escalation 3% per y	Subtotal Contingency Subtotal ear to 2004 Subtotal	316,564 3,648,816 182,441 3,831,257 95,781 3,927,039 392,704 4,319,743 129,592 4,449,335 133,480 4,582,815

Appendix C

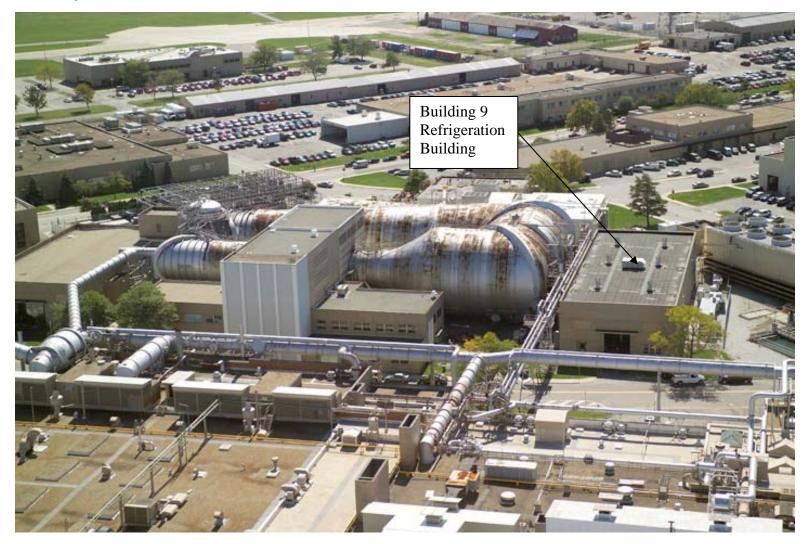
<u>Ariel Photographs</u> <u>Of the</u> <u>Altitude Wind Tunnel</u>



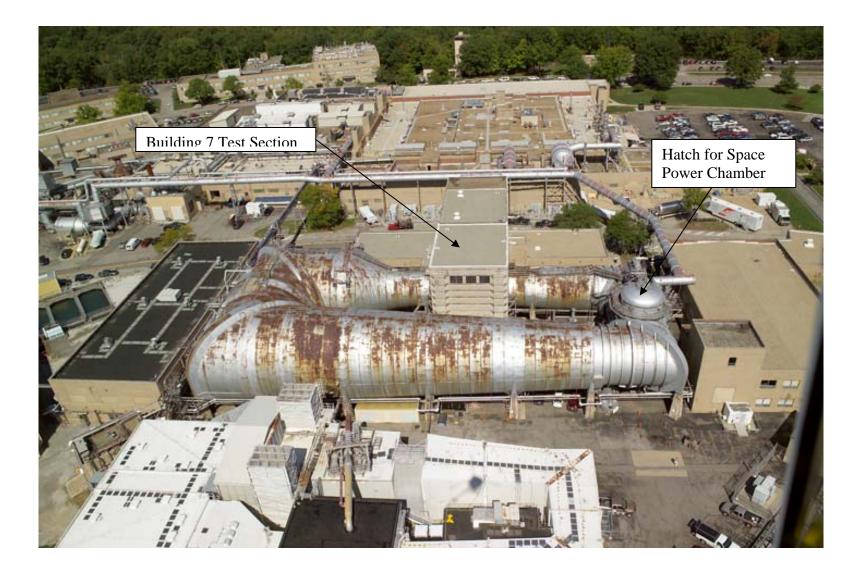
Photograph No. 1 - Altitude Wind Tunnel Viewed from the South with the Icing Research Tunnel and Substation "B" in the Foreground



Photograph No. 2 – Altitude Wind Tunnel Viewed from the Northeast with the Drive Motors and Compressor Building to the Left.



Photograph No. 3 – Altitude Wind Tunnel Viewed from the Northwest with the Refrigeration Building to the Right.



Photograph No. 4 – Altitude Wind Tunnel from the South, Test Section in the Building in the Center of the Picture.